REPORT DOCUMENTATION PAGE

Form Approved OMB No. 0704-0188

Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing this collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden to Department of Defense, Washington Headquarters Services, Directorate for Information Operations and Reports (0704-0188), 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302. Respondents should be aware that notwithstanding any other provision of law, no person shall be subject to any penalty for failing to comply with a collection of information if it does not display a currently valid OMB control number. PLEASE DO NOT RETURN YOUR FORM TO THE ABOVE ADDRESS.

1. REPORT DATE (DD-MM-YYYY)	2. REPORT TYPE	3. DATES COVERED (From - To)		
17 September 2015	Briefing Charts	24 August 2015 – 17 September 2015		
4. TITLE AND SUBTITLE		5a. CONTRACT NUMBER		
Verification of DSMC Simulations Usi	ng Spectral Analysis			
		5b. GRANT NUMBER		
		5c. PROGRAM ELEMENT NUMBER		
6. AUTHOR(S)		5d. PROJECT NUMBER		
S. J. Araki and R. S. Martin				
		5e. TASK NUMBER		
		5f. WORK UNIT NUMBER		
		O0A5		
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES)	8. PERFORMING ORGANIZATION REPORT NO.		
Air Force Research Laboratory (AFMC	C)			
AFRL/RQRS				
1 Ara Drive				
Edwards AFB, CA 93524-7013				
9. SPONSORING / MONITORING AGENCY	NAME(S) AND ADDRESS(ES)	10. SPONSOR/MONITOR'S ACRONYM(S)		
Air Force Research Laboratory (AFMC	C)			
AFRL/RQR		11. SPONSOR/MONITOR'S REPORT		
5 Pollux Drive		NUMBER(S)		
Edwards AFB, CA 93524-7048		AFRL-RQ-ED-VG-2015-349		
12. DISTRIBUTION / AVAILABILITY STAT	EMENT	1		

Approved for public release; distribution unlimited

13. SUPPLEMENTARY NOTES

For presentation at DSMC15; Kapaa, Kauai, Hawaii; 13-17 Sep 2015

PA Case Number: #15556; Clearance Date: 9/15/2015

14. ABSTRACT

Viewgraphs/Briefing Charts

15. SUBJECT TERMS

N/A

16. SECURITY CLAS	SIFICATION OF:		17. LIMITATION OF ABSTRACT	18. NUMBER OF PAGES	19a. NAME OF RESPONSIBLE PERSON J. Koo
a. REPORT	b. ABSTRACT	c. THIS PAGE	SAR	18	19b. TELEPHONE NO (include area code)
Unclassified	Unclassified	Unclassified			N/A



Verification of DSMC Simulations Using Spectral Analysis

SAMUEL J. ARAKI AND ROBERT S. MARTIN

ERC INC, AIR FORCE RESEARCH LABORATORY EDWARDS AIR FORCE BASE, CA USA





DISTRIBUTION A: APPROVED FOR PUBLIC RELEASE; DISTRIBUTION UNLIMITED AFTC/PA CLEARANCE NO. TBD





Outline



• Introduction

• Test Problem

• Chi-Square Test

Spectral Analysis

Conclusion / Future Work

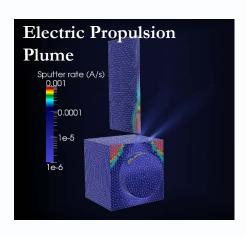


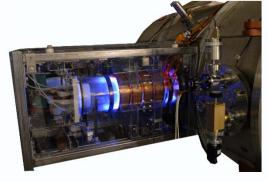
TURF:



Thermodynamics Universal Research Framework

- Unifies multiple research codes from group
 - Models of spacecraft propulsion relevant plasma at different spatial scales (1μm-100m)
 - Single fluid Magnetohydrodynamics (MHD) solver
 - Unstructured multi-fluid solver
 - Vlasov solver
 - Particle-In-Cell (PIC) with DSMC
 - Combined redundant functionality in those codes
 - Unified GPU support





Field Reversed Configuration Plasma





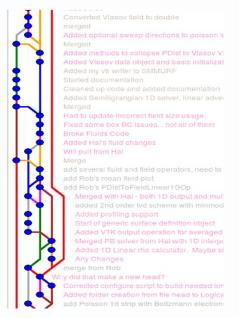
Motivation / Objectives



Motivation

- Multiple developers of TURF
 - Changes to core class objects required by one model can break the others.
 - Merging different heads and making sure all the models work properly
- Automated test suites for different models

Multiple heads in Mercurial



Objectives

- Set up simple problem for testing DSMC part of TURF
- Development of efficient procedure in verifying the results



Test Problem: 1D Normal Shock



- Standalone DSMC code run on GPU
- Initial State
 - Upstream and downstream flow properties according to Rankine-Hugoniot relations
 - Discontinuous profile at t=0

$$M_{2}^{2} = \frac{1 + \frac{\gamma - 1}{2} M_{1}^{2}}{\gamma M_{1}^{2} - \frac{\gamma - 1}{2}},$$

$$\frac{n_{2}}{n_{1}} = \frac{u_{1}}{u_{2}} = \frac{(\gamma + 1) M_{1}^{2}}{(\gamma - 1) M_{1}^{2} + 2},$$

$$\frac{T_{2}}{T_{1}} = 1 + \frac{2(\gamma - 1)}{(\gamma + 1)^{2}} \frac{\gamma M_{1}^{2} + 1}{M_{1}^{2}} (M_{1}^{2} - 1)$$

	Upstream ₁	Downstream ₂
Mach Number, M	1.20	0.85
Density, n (m ⁻³)	1.00×10^{22}	1.30×10^{22}
Temperature, T (K)	293.0	350.1



Test Problem: 1D Normal Shock

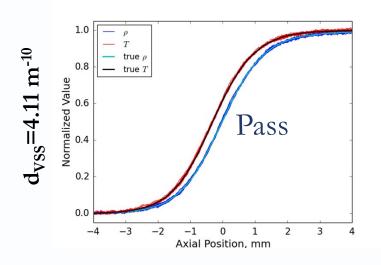


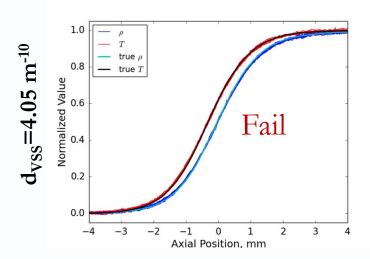
Requirements

- Fast simulation (at least < 5 hours) less particles
- Smooth solution more particles and time-steps

Baseline inputs: (40 mins)

• 8,000,000 Ar particles, 10,000 time-steps, 1,000 sampling cells





How can we systematically accept/reject results?



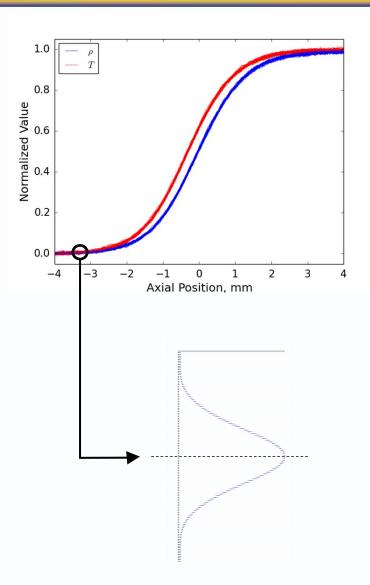
Chi-Square Test: Normal Distribution



1. Run the test problem *M* times with different random number seed

2. Deviation of results is expected to fall under normal distribution

$$f(x,\mu,\sigma) = \frac{1}{\sigma\sqrt{2\pi}} \exp\left(-\frac{(x-\mu)^2}{2\sigma^2}\right)$$



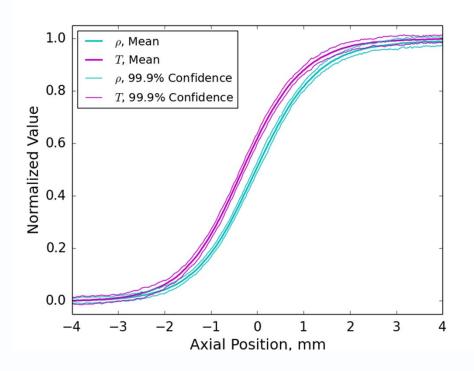


Chi-Square Test: Standard Deviation



3. Determine point-wise expected value (μ) and standard deviation (σ) for N cells

$$\mu_n = \frac{1}{M} \sum_{m=1}^{M} x_{n,m}, \quad \sigma_n = \frac{1}{M-1} \sum_{m=1}^{M} (x_{n,m} - \mu_n)^2$$



% Confidence Level	# of σ s from Mean
99.9	3.2905
99	2.5758
95	1.9600
90	1.6449
80	1.2816
70	1.0364
50	0.6745



Chi-Square Test



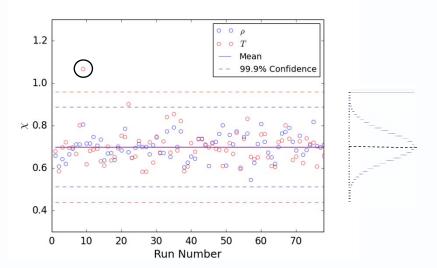
4. For a given run, deduce a global parameter (χ) using μ and σ

$$P = \left[\prod_{n=1}^{N} f_n \sigma_n \sqrt{2\pi}\right]^{1/N} = \prod_{n=1}^{N} \exp\left(-\frac{(x_n - \mu_n)^2}{2N\sigma_n^2}\right)$$

$$\Rightarrow -\log P = \frac{1}{2N} \sum_{n=1}^{N} \left(-\frac{(x_n - \mu_n)^2}{\sigma_n^2}\right) = \chi^2 < \chi_{\text{ref}}^2$$

VSS Diameter	χ		
(Angstrom)	9	Т	
4.11	0.753	1.02	
4.00	1.10	1.06	
3.90	1.33	1.31	
3.80	2.79	2.64	

Generally increasing



^{*} All runs performed with the same RNG seed



Chi-Square Test: Result



- Plots of two normal distributions for different VSS diameter
 - The mean of one distribution is larger than the other with larger cross-section, but some portion of it should still overlap

Distribution overlaps...

Could we effectively identify the distribution given a value in the overlapping region?



Remove/reduce statistical noise

Run more particles and time-steps?

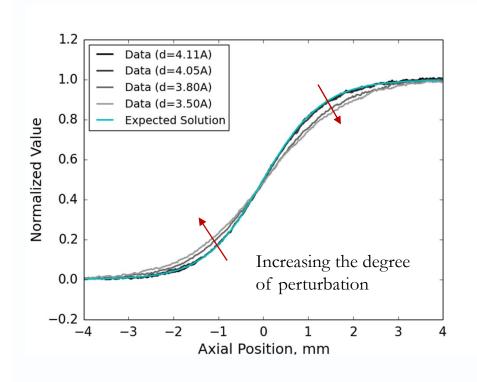


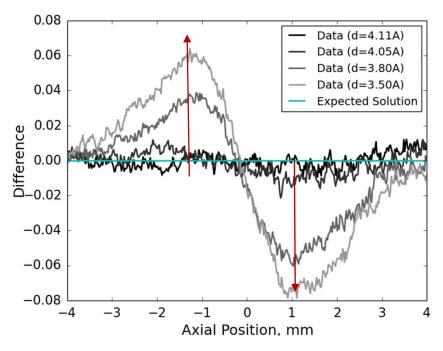
Spectral Analysis: Difference Profiles



Perform analysis in frequency space

- High frequency mode Purely statistical
- Low frequency mode Comes from perturbed input

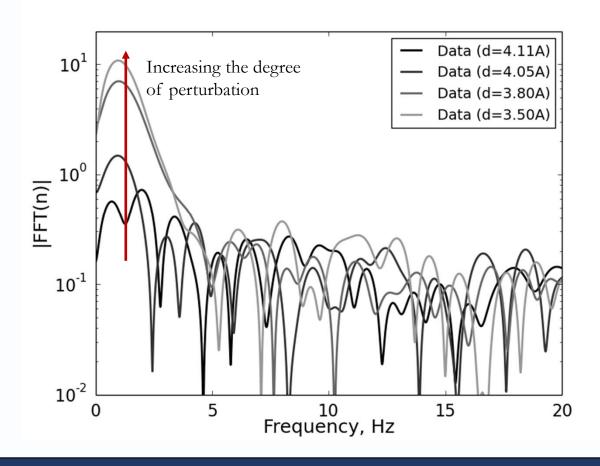






Spectral Analysis: FFT on Difference Profile



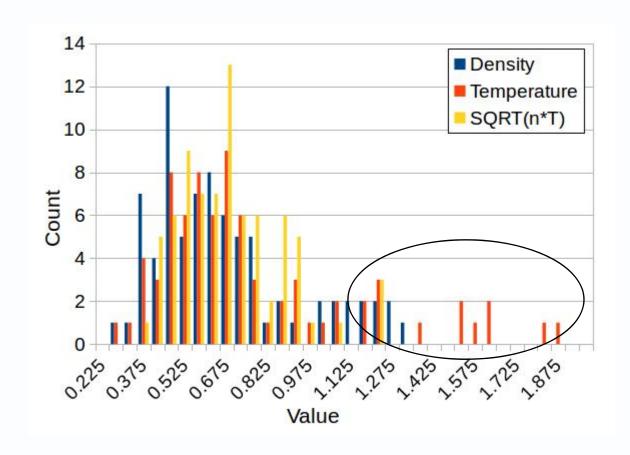


Smaller low mode amplitude as approaches the true input solution



Spectral Analysis: Distribution of Low-Mode Amplitude





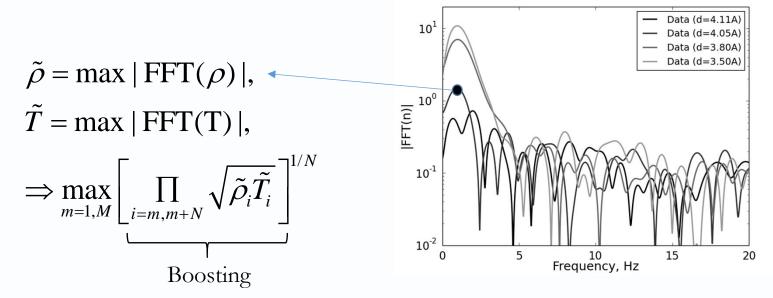
How do we deal with outliers...?



Spectral Analysis: Boosting



- Need to pass the runs with outliers
 - What is the probability of outliers occurring multiple times on the roll?
 - Probability of one outlier: 1%
 - Probability of outliers three times on the roll: $(1\%)^3$
- Use geometric mean to obtain the global spectral amplitude





Conclusion / Future Work



Conclusions

• Chi-Square test

- Width of distribution is determined by the statistical noise inherent to particle simulations

• Spectral analysis

- Strongly effective in the test problem
- Essentially removes the effect of statistical noise in the test

Future Work

- Use the methodology in other DSMC problems to study the strength/weakness of the procedure
- Apply the knowledge in smoothing of flow properties





Thank you



Spectral Analysis



- 1. Perturbing the VSS diameter narrows/widens the shock width
- 2. Difference profile contains
 - 1. Low frequency mode from wrong input
 - 2. High frequency mode from statistical noise
- 3. Perform Fast Fourier Transformation (FFT) on difference profile
- 4. Observe the amplitude of low modes
- 5. Perform boosting